# AD-A258 568



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330 West 42nd Street / New York, New York 10036 / (212) 563-4545

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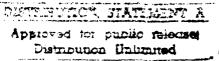
**TECHNICAL REPORT TR-1/C68-603B** 

TREND FIELD TEST REPORT

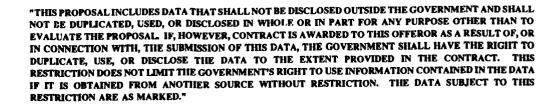


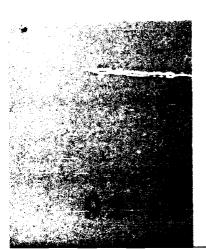
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Prepared for: Naval Air Warfare Center Aircraft Divison P.O. Box 5152 Warminster, PA 18974-0591 Att: Mr. Tom Pohle, 5011/Mr. Mike Hess, 5014





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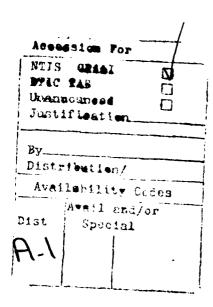
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#### INTRODUCTION

The following report describes work carried out, leading up to and including the final ground based field test of the Testbed for Realistic Evaluation of Novel Discriminants (TREND) electro-optical system. The ultimate goal of this system is to provide a means of passive discrimination between moving objects and a stationary cluttered background by virtue of signal source angular velocity separation. Angular velocity may be inferred by signal frequency on a single periodically masked detector as the source moves across its field of view. The signal amplitude must vary sufficiently over an angular subtense equal to half the mask period to register above the local noise. Such signals may arise from point objects (with respect to half the mask period) or hard edges perpendicular to the mask period.

An additional twist to this method, utilized in the TREND sensor, optically filters selected spatial frequencies by dividing the wavefront at the aperture, transmitting radiation through a double slit. If spatial coherence is maintained over the entrance aperture a periodic interference pattern is formed which when properly matched to the mask period will exhibit a periodic signal similar to that produced with an undivided wavefront. The advantage to this scheme is that spatial filtering is performed optically. The obvious disadvantage is reduction of received energy. The current TREND system uses a scanning sensor to simulate moving radiation sources with an actually stationary background, thus the background source signals will be modulated at the nominal frequency proportional to the scanning angular velocity.

The ground based field test effectively demonstrated the theoretical signal modulation (under controlled conditions), yet system noise and lack of detector sensitivity prohibited the collection of sufficiently strong signals from the natural environment. The field testing was conducted at the Naval Air Warfare Center (NAWC) in Warminster, PA from June 22, 1992 to June 24, 1992. The sensor was mounted in a third story room, scanning horizontally over a parking lot and airfield (appendix A, figure 3). The design periodic signal was observed while scanning an artificial blackbody source at 206 meters; no such signal was observed while scanning a welding torch at 1536 meters.

#### DOUBLE SLIT INDUCED INTERFERENCE

The novelty of the TREND sensor stems in its use wavefront division to alter the MTF of the optical system such that it acts as a spatial filter. The detector mask is used to translate angular velocity information to the time domain for display, recording, and discrimination purposes.

A useful interferometric model treats the TREND aperture as the classic Young's double slit illuminated by a plane wave. Interference of the wavefronts produces an intensity distribution proportional to the cosine of the phase difference between each wavefront. As seen in figure 1,

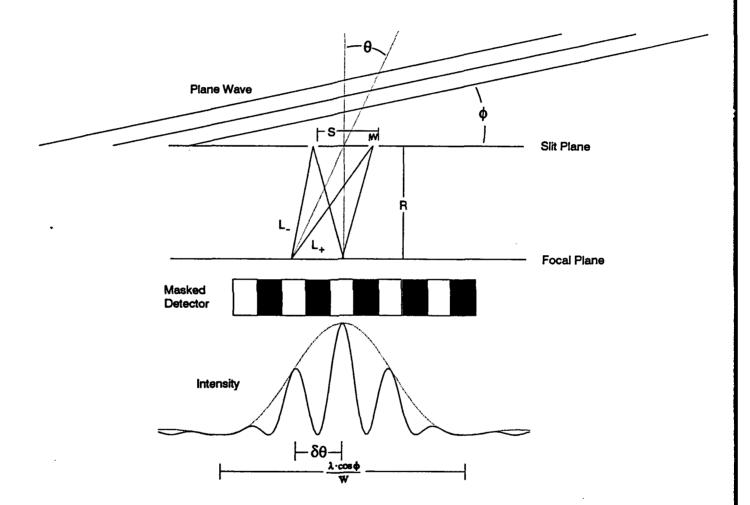


Figure 1. Fringe Model

the angular separation,  $\delta\theta$ , between fringes is  $\lambda/s \cdot \cos\phi/\cos\theta \cdot \sqrt{1 + (s/2R)^2}$ , where  $\lambda$  is the wavelength of light, s is the center to center separation of the slits,  $\theta$  is the fringe observation angle with respect to the slit plane,  $\phi$  is the plane wave incident angle with respect to the slit plane, and R is the range at which the pattern is observed (in our case the focal length of the imaging optics). Note that the entire focal length of the system is not used since the wavefront is divided between the afocal telescopic optics and imaging optics. The schematic figure 1 shows the fringe pattern modulated by the point spread function of a single slit of width w. Most of the energy is contained within the angular subtense  $\lambda/w\cos\phi$ . Figure 1 also shows a portion of a single detector masked at the appropriate spatial frequency to give maximum signal modulation as the fringe pattern moves across its field of view. The modulation occurs at a frequency equal to the source angular velocity divided by the detector mask angular subtense.

The following provides a simple model for fringe separation with respect to design parameters. Referring to figure 1 we see that the path difference, PD, from the center of each slit to the focal plane is given by

$$PD = L_{-} - L_{-}.$$

When  $\frac{\partial PD}{\partial \theta} \cdot \delta \theta = \lambda \cdot \cos \phi$  consecutive intensity maxima are observed.

Since

$$L_{\perp} = R\sqrt{1 + (\sin\theta \pm s/2R)^2}$$

$$\frac{\partial L_{\pm}}{\partial \theta} = \frac{R \cos \theta (\sin \theta \pm s/2R)}{\sqrt{1 + (\sin \theta \pm s/2R)^2}}$$

therefore

$$\frac{\partial PD}{\partial \theta} = \frac{R\cos\theta(\sin\theta + s/2R)}{\sqrt{1 + (\sin\theta + s/2R)^2}} - \frac{R\cos\theta(\sin\theta - s/2R)}{\sqrt{1 + (\sin\theta - s/2R)^2}}$$

Close to the center of the field  $\sin \theta \ll s/2R$ , so

$$\frac{\partial PD}{\partial \theta} = \frac{s \cdot \cos \theta}{\sqrt{1 + (s/2R)^2}}$$

and

$$\delta\theta = \frac{\lambda \cdot \cos\phi}{\frac{\partial PD}{\partial \theta}} = \frac{\lambda \cdot \cos\phi\sqrt{1 + (s/2R)^2}}{s \cdot \cos\theta}$$

Available fringe contrast may be maximized by minimizing the ratio of the sum of the partial differentials of the fringe separation with respect to each design parameter (i.e.,  $\lambda$ , s,  $\phi$ , w, f1) to the fringe separation. Stated mathematically the relative fringe shift (RFS) is,

RFS = 
$$\sum_{i=1}^{n} \frac{\partial \delta \theta}{\partial x_{i}} \cdot \frac{\delta x_{i}}{\delta \theta}$$

where the elements, i, are given as follows:

$$\lambda - \frac{\delta \lambda}{\lambda}$$

$$s - \frac{-\delta s}{s[1 + (\frac{s}{2R})^2]}$$

$$\dot{\Phi} = \frac{\delta \phi \tan \phi}{1 + (\frac{s}{2R})^2}$$

$$R = \frac{-(\frac{s}{2R})^2 \delta R}{R[1 + (\frac{s}{2R})^2]}$$

Note that  $\delta\lambda$  is limited by the optical filter bandwidth,  $\delta s$  is limited to the slit width,  $\delta \phi$  represents a shift in the effective plane wave incident angle and  $\delta R$  represents a shift in the detector plane from the design focal plane.

### ATMOSPHERIC TURBULENCE

As mentioned previously, interference fringes will be produced if the wavefront approximates a plane wave at the entrance aperture. This coherent area is usually quantified in terms of the coherence diameter,  $r_0$ . This is a function of atmospheric turbulence, the path length over which the radiation travels, and the wavelength of light. It is given by formula as follows:

$$r_0 = (0.4233 C_n^2 k^2 L)^{-3/5}$$

where:

 $C_n^2$  = refractive index structure parameter

k = radiation wavenumber

L = path length.

We did not measure the coherence diameter during the field test, but may estimate it based upon typical values of  $C_n^2$  [1]. Close to the ground, the refractive index structure parameter is highly dependent upon the time of day and type of surface over which it is

measured. Generally the value is proportional to the degree of thermal equilibrium between the surface and the atmosphere. Using bounding values of  $C_n^2$  one may estimate reasonable coherence diameters. These are shown for various ranges in table 1.

Table 1

	$r_0$ (m) for $C_n^2$ (m <sup>-2/3</sup> )				
L (m)	1.5 x 10 <sup>-12</sup>	1.5 x 10 <sup>-14</sup>			
1	21	330			
100	1.3	21			
206	0.90	14			
1000	0.34	5.3			
1536	0.26	4.0			

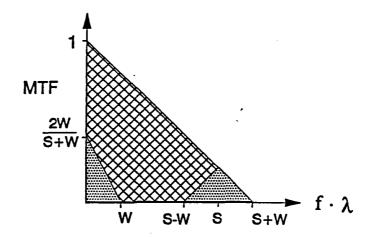
### **OPTICAL MTF**

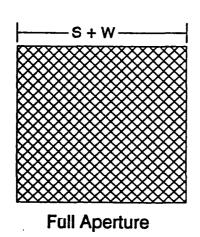
The aperture is divided to alter the MTF (modulation transfer function) such that undesired spatial frequencies are filtered out. In general, the MTF of an optical system (in cycles/radian) is the autocorrelation function of the entrance pupil divided by the admitted radiation wavelength. The slit pupil patterns used in the TREND system increase the high to low frequency MTF ratio with respect to that of a non-divided pupil, while maintaining the same absolute MTF at high frequency. A simple comparison of full aperture and slit aperture MTFs can be made by approximating the apertures as rectangular. These theoretical MTFs are graphed in figure 2, normalized to the full aperture. The actual TREND MTFs will differ slightly because those shown assume rectangular pupils and diffraction limited optics. The signal available at any frequency is proportional to the product of the admitted energy and the optical MTF. Given equal irradiances at the entrance aperture, the admitted energy is proportional to the pupil area. At spatial frequencies greater than the design frequency ( $f = s/\lambda$ ) the slit pupil signal will be modified by the pupil area reduction, while at lower frequencies, modified by the product of the MTF and pupil area reductions.

### PRELIMINARY LABORATORY TESTING1

Prior to performing the ground based field test the system was tested in the laboratory with a soldering iron source and vertical slit aperture measuring  $5 \times 10^{-3}$  by  $2.5 \times 10^{-5}$  radians.

All laboratory testing performed at chip carrier temperature equal to 80 K. Field testing was performed at 70 K.





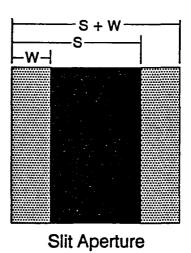


Figure 2. Theoretical MTF

The iron temperature was 593 K, measured with a thermocouple. Preamplifiers were set at 10<sup>6</sup> volts/amp gain.<sup>2</sup> With the scanner stationary the radiation was chopped at 20 Hz just prior to the slit pupil. Signal and noise data was recorded for various pupil and detector element combinations as shown in table 2.

Table 2

FILTER 2		PUPIL 2				
PUPIL	SIGNAL (mV)	FILTER	SIGNAL (mV)	NOISE (mV @ 120 Hz)		
1	80	0	120	20		
2	100	1	10	20		
3	160	2	100	20		
4	580	3	10	20		
5	580	4	55	20		
6	580	5		100		
		6	20	40		
		7	20	40		

Additionally, with pupil 2 (appendix A, figure 2) inserted, the detectors were scanned across the source at five speeds ranging from  $2 \times 10^{-3}$  to  $2 \times 10^{-4}$  radians/second. Data was digitized and recorded on disk from eight multiplexed channels. A list of those data files is shown in appendix B.

### **NOISE CONSIDERATIONS**

In general the system was found to be very noisy, predominated by the fundamental 60 Hz line frequency and multiples. Testing in the laboratory using the dual trace oscilloscope as a differential amplifier proved that the noise was common to both the detector ground and signal leads. Inputting a detector signal lead directly to the oscilloscope without grounding the detector to chassis gave a 1.5 volt signal at 60 Hz. Grounding the detector to chassis reduced this to 5 millivolts random frequency. This was analogous to the normal operating condition, except that the oscilloscope provided voltage to voltage amplification while the preamplifiers provided current to voltage amplification. Under normal operating conditions noise was reduced to a minimum of 20 millivolts at 120 Hz after passing through the Texas Instruments (TI) filter circuit (low pass 60 Hz notch filter, -3 dB at 20 Hz). More typical noise values ranged up to 100 millivolts. The greatest noise reduction was obtained by bringing the detector ground and signal

Over the course of the field testing the preamplifiers gains were set at 10<sup>5</sup> or 10<sup>6</sup> volts/amp since system noise would cause saturation at higher values.

leads to two channels of the oscilloscope, then inverting one channel. This differential amplifier configuration gave a random output voltage of 2 millivolts.

Generally each preamplifier showed a DC offset around 400 millivolts, the exception being number 10 with an offset of 70 millivolts. Preamplifier 10 outputs were the noisiest.

Common mode noise predominates in the TREND system because the preamplifiers use single ended inputs; the non-inverting input is carried by the coaxial signal line shield (appendix A, figure 1). Due to multiple grounding of the shield (at both the dewar and amplifier box) a current flows between these points which couples with the real signal current. The problem is further magnified by capacitive coupling between 22 unused detector lead coaxial cables and a noisy laboratory environment since the shields are all connected to the dewar ground point [3]. This is evidenced by noise changing by an order of magnitude as a person moves around the room or adjusts the cable positions. Scanning operation gives rise to continually changing noise values. Proper grounding and shielding employs biaxial or triaxial cables to bring the signal ground and detector leads to the amplifiers, and grounding of the entire shield (dewar, cable, and preamplifier box) only at the dewar [4]. The signal ground lead should be attached between the dewar ground point and the preamplifier non-inverting input. The detector output should be attached to the preamplifier inverting input. Ideally each detector should have both output and ground leads; however, the TREND chip carrier uses a common ground for all 32 detectors. This common signal ground should be buffered before splitting to the non-inverting inputs of each preamplifiers [5].

A further refinement reduces the effect of common mode noise produced by leakage through the shield by placing a resister of equal magnitude to that of the feedback resister in series with the non-inverting amplifier input and signal ground point [6].

#### FIELD TEST

The ground based field test was conducted from a third story room in the NAWC facility with the sensor 14 meters above the ground scanning at 1.57 x 10<sup>-3</sup> radians/second out toward the eastern horizon. Unobscured line of sight scan could be obtained for -45 to +45 degrees in azimuth and -10 to +30 degrees in elevation. Due to system noise the preamplifiers saturated at high gain so full detector sensitivity was unavailable. An artificial blackbody source of 1.75 x 10<sup>-2</sup> meter diameter at 206 meters range was scanned successfully at temperatures 375 C, 500 C, and 600 C. A welding torch (assumed temperature 1700 C) at 1536 meters range did not produce an observable signal (see Energy Considerations for justification). A map indicating sensor and source positions is shown in figure 3. We observed one case of frequency modulation with a full aperture pupil when the sensor was not scanning. We assume sensor or turbulence induced wavefront jitter caused these signals.

#### **ENERGY CONSIDERATIONS**

The noise equivalent input (NEI) of the system is the contrast irradiance at the TREND aperture which will produce a signal to noise ratio of unity. This is used to quantify sensitivity.

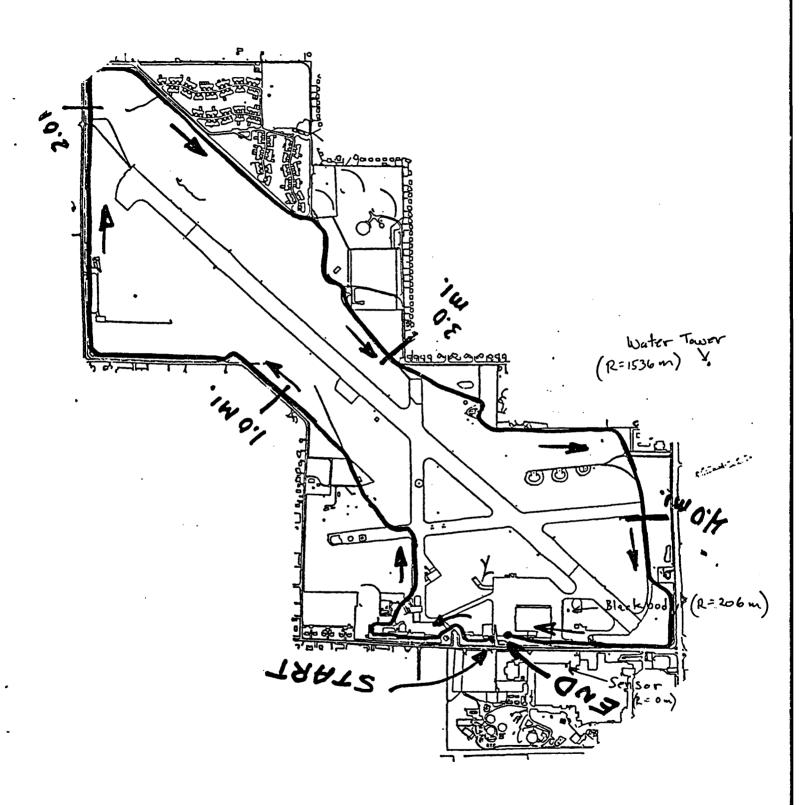


Figure 3. Source - Sensor Geometry

For a subpixel source it may be calculated as follows:

Δλ

$$CE(A,R,T,\Delta \lambda) = \frac{A \cdot \tau(R) \cdot [L(\Delta \lambda, T_{source}) - L(\Delta \lambda, T_{back})]}{R^2}$$
where:
$$CE = \text{contrast irradiance}$$

$$A = \text{area of the source}$$

$$R = \text{range to the source}$$

$$\tau(R) = \text{atmospheric transmission over range } R$$

$$L = \text{radiance}$$

$$T = \text{absolute temperature}$$

We calculated the contrast irradiance at the sensor for three different blackbody source temperatures. The atmospheric transmission was assumed to be unity over the dewar window spectral transmission band (1 micron width centered on 10 microns) [6]. Voltage measurements were taken via oscilloscope from TI channel 3 with preamplifier gain set to 10<sup>6</sup> volts/amp and chopping frequency at 25 Hz. Table 3 shows the contrast irradiances, signal voltages, and resultant noise equivalent inputs. In all cases the noise was 100 millivolts.

waveband

Table 3

TEMPERATURE (K)	CE (w/m²)	SIGNAL (V)	NEI (w/m²)
873	9.27 x 10 <sup>-7</sup>	1.50	6.17 x 10 <sup>-8</sup>
773	7.08 x 10 <sup>-7</sup>	1.25	5.65 x 10 <sup>-8</sup>
648	4.58 x 10 <sup>-7</sup>	0.76	6.02 x 10 <sup>-8</sup>

Note that these NEIs were measured with a full aperture pupil. Slit pupil apertures increase these values by factors ranging from 2.6 to 5.0 (appendix A, figure 2).

Since we did not observe the expected modulated signal from the welding torch<sup>3</sup> (a very hot source), we calculated its contrast irradiance at the sensor. The resultant value  $(1.7 \times 10^{-7} \text{ w/m}^2)$  is close enough to the NEI to justify the experimental results.

Assumptions - hot area = 4 cm<sup>2</sup>, temperature = 1973 K [7], atmospheric transmission = 1 (worst case 0.5) [8].

### **REFERENCES**

- [1] W. Wolfe, G. Zissis, "The Infrared Handbook," Environmental Research Institute of Michigan, (1985) pp. 6-11 6-14.
- [2] R. Morrison, "Grounding and Shielding Techniques in Instrumentation," 3rd Edition, Wiley, New York, (1986) p. 145, 148.
- [3] P. Horowitz, W. Hill, "The Art of Electronics," Cambridge University Press, Cambridge, (1980) p. 308.
- [4] Ref. [2] pp. 37 54.
- [5] Ref. [3] p. 311.
- [6] Ref. [3] p. 113.
- [7] D. Lide, "CRC Handbook of Chemistry and Physics," 72nd Edition, CRC Press, Boca Raton, (1991) p. 12-136.
- [8] "Electro-Optics Handbook," 2nd Edition, RCA Commercial Engineering, Harrison, NJ, (1974) pp. 98 100.

APPENDIX A

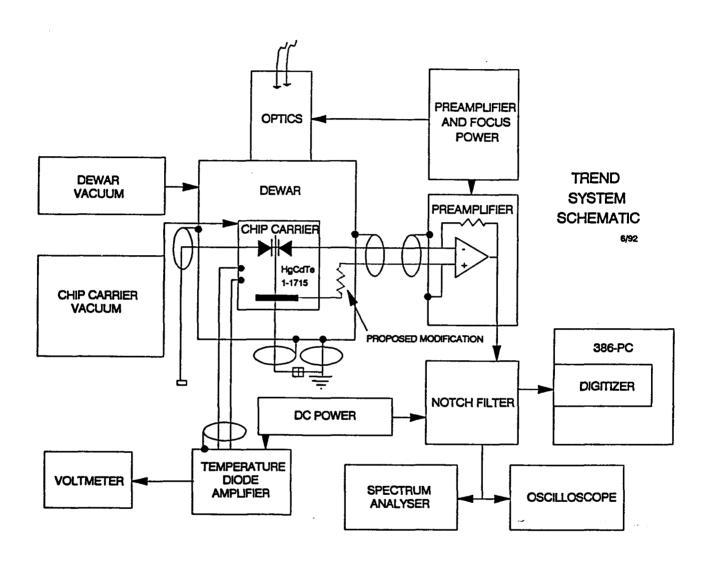
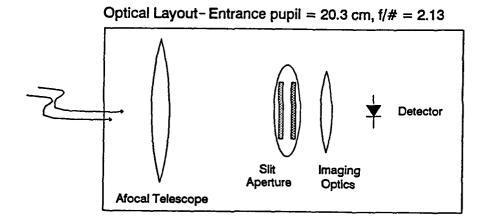
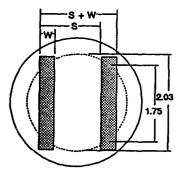


Figure 1. System Schematic





# Pupil Dimensions (cm)

Pupil	S	W	S/W	R
		0.183		
		0.239		
3	1.044	0.348	3.00	0.38

R = slit pupil area / circular pupil area

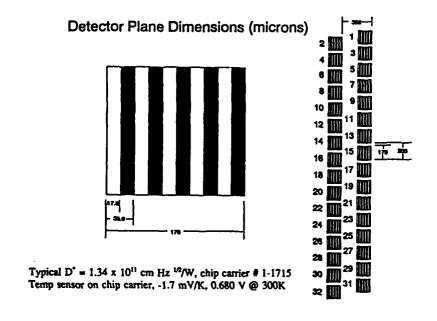


Figure 2. Optical Dimensions

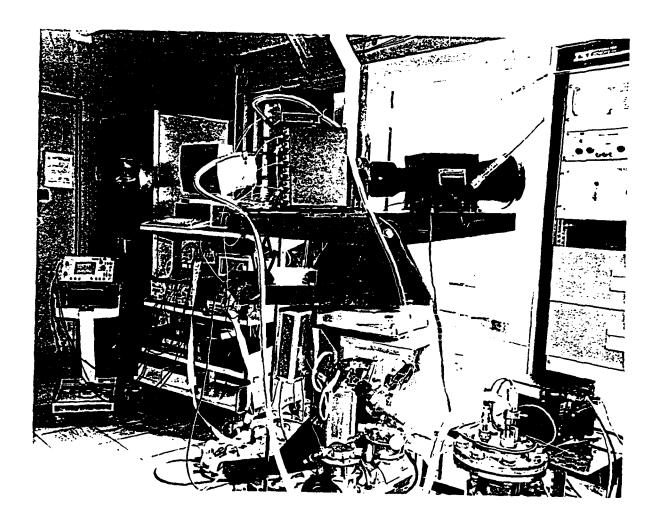


Figure 3. Photograph of TREND System

APPENDIX B

### **DATA COLLECTION**

The spectrum analyzer plots represent the output of TI channel 3, lowpass filtered by the spectrum analyzer with a cutoff at 100 Hz.

The digitized data files represent 15 seconds of raw data from TI channels 0, 1, 2, 3, 4, 5, 6, 7, time sampled at 128 Hz. The 4 seconds shown on the spectrum analyzer plots are located somewhere within this time frame, typically after the 5 seconds of pretrigger data.

The following Labtech Notebook icon-generated software block files drive the Metrabyte DASH-16 digitizer board:

- a. TRIG8 Digitize, time stamp, display, store channels 0 7.
- b. SCANODD Digitize, display channels 1, 3, 5, 7.
- c. SCANEVEN Digitize, display channels 0, 2, 4, 6.
- d. SCAN03 Digitize, display channels 0, 1, 2, 3.
- e. SCAN47 Digitize, display channels 4, 5, 6, 7.
- f. AVETEST Replay channel 0 with various averaging schemes (0 10 sec).
- g. EVEN Replay channels 0, 2, 4, 6 (0 10 sec).
- h. ODD Replay channels 1, 3, 5, 7 (0 10 sec).
- i. REPLAY3 Replay channel 3 (0 15 sec).

# **DIGITIZED DATA FILES**

base#, basx# - 6/22/92

basy# - 6/23/92 basz# - 6/24/92

FILE	PUPIL	GAIN (V/A)	SCAN (deg/s)	COMMENT
basel	6	105	0.038	water tower/welder
base2	2			water tower/no welder
basx0				water tower/no welder
basx1				trees at 1 km
basx2				aperture covered
basx3				aperture covered
basx4				water tower/welder
basx5		-		water tower/welder
basx6				water tower/welder
basx7				water tower/welder
basx8		-		water tower/welder
basx9				water tower/welder
basx10		10 <sup>6</sup>		water tower/welder
basx11			0.025	water tower/welder
basx12			0.019	water tower/welder
basx13				water tower/welder
basx14				crane cable at water tower
basx15				crane cable at water tower
basx16				cloud along horizontal edge
basx17				cloud along vertical edge
basx18				cloud along vertical edge
basx19				nearby cloud
basx20				tree tops at 1 km

FILE	PUPIL	GAIN (V/A)	SCAN (deg/s)	COMMENT
basx21	2	10 <sup>6</sup>	0.093	water tower/welder
basx22				water tower/no welder
basx23				aperture covered partially
basx24				aperture covered
basx25			-	aperture covered, person moving
basx26				aperture covered then uncovered
basx27				clouds
basx28			0.087	crane cable at water tower
basx29				water tower/welder
basx30				water tower/no welder, trees
basx31				plane w/props moving at 200 m
basx32				tarmac at 200 m
basx33				?
basx34			?	?
basx35				water tower/no welder
basx36				water tower/no welder
basx37			_	water tower/welder
basy0	2	10 <sup>6</sup>	?	?
basy1			0.000	aperture covered
basy2				aperture open
basy3				aperture covered
basy4			0.093	crane cable and crane at tower
basy5				tree tops at 1 km
basy6				?
basy7				?
basy8				?
basy9				?

FILE	PUPIL	GAIN (V/A)	SCAN (deg/s)	COMMENT
basy10	2	10 <sup>6</sup>	0.093	?
basy11				?
basy12	4			1.35 cm diameter, 600 C BB @ 206 m
basy13	2			1.35 cm diameter, 600 C BB @ 206 m
basy14				1.35 cm diameter, 600 C BB @ 206 m
basy15				?
basy16	3			1.35 cm diameter, 600 C BB @ 206 m
basy17	2			1.35 cm diameter, 600 C BB @ 206 m
basy18				1.35 cm diameter, 600 C BB @ 206 m
basy19	1			1.35 cm diameter, 600 C BB @ 206 m
basy20				1.35 cm diameter, 600 C BB @ 206 m
basy21	3			1.35 cm diameter, 500 C BB @ 206 m
basy22	4			1.35 cm diameter, 500 C BB @ 206 m
basy23				1.35 cm diameter, 500 C BB @ 206 m
basy24			0.000	grass background at 206 m
basy25				grass background at 206 m
basy26			0.093	crane wire at tower
basy27			0.093	1.35 cm diameter, 375 C BB @ 206 m
basy28				?
basy29	3		0.093	1.35 cm diameter, 375 C BB @ 206 m
basy30				1.35 cm diameter, 375 C BB @ 206 m
basy31				1.35 cm diameter, 375 C BB @ 206 m
basy32				clouds
basz0	2		0.000	covered aperture, typical noise
basz1				vered, typical spikes
basz2			0.093	building at about 200 m
basz3				covered aperture

FILE	PUPIL	GAIN (V/A)	SCAN (deg/s)	COMMENT
basz4	2	10 <sup>6</sup>	0.093	uncommonly noisy scan
basz5				uncommonly noisless scan
basz6				cumulous clouds, 1-2 km
basz7				30 Hz signal when scanner -20 degrees azimuth, cumulous clouds 1-2 km range, repeatable w.r.t. scanner position, not cloud position
basz8-20				not noteworthy
scan0			6.123	
scan1			0.048	
scan2			0.031	320 C, 5 x 10 <sup>-3</sup> by 2.5 x 10 <sup>-5</sup> radian
scan3			0.019	source at focal point of 3.05 meter collimater
scan4			0.014	

APPENDIX C

# <u>DATA-FILE - SPECTRUM ANALYZER PLOT CROSS REFERENCE</u>

Spectrum Analyzer Plot	Digitized Data File
1	basy12
2	basy30
3	basy31
4	basy32
5	basy24
6	basy27
7	basy29
8	basy21
9	basy18
10	typical of basy19,20
11	basy17
12	basy16
13	none, typical transient with
	aperture covered
14	basy23
15	basy25
16	none, typical noise spike
17	typical of basz5, 6
18	basz7
19	basz4
20	none, relatively low noise
21	basz5 ?
22	basz6 ?
23	basz5 ?
24	basz4
25	basz0
26	basz1

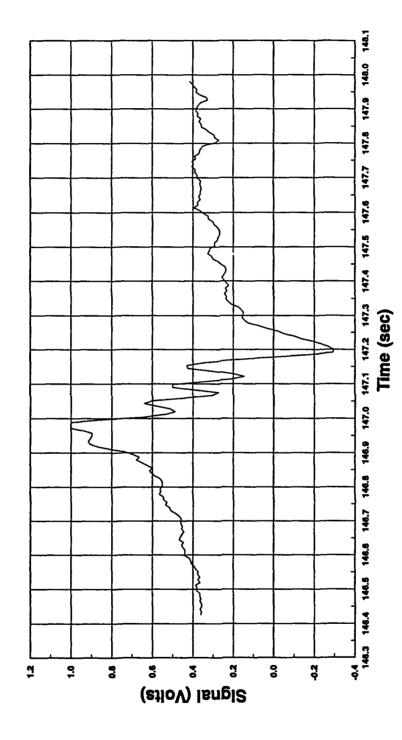
WIRING CROSS REFERENCE #'S								
TI FILTER	0	1	2	3	4	5	6	7 OF 0-7
ARRAY	15	16	17	18	19	20	21	22 OF 32
NERC PIXEL	5	6	7	8	9	10	11	12 OF 16
CHIP CARRIER	8	44	9	43	10	42	11	41 OF 68
BNC	9	28	10	27	11	26	12	25 OF 36
PREAMP	2	3	6	7	9	10	14	15 OF 16
37D CONNECTOR	36	35	32	31	18	17	13	12 OF 37
D/A CHANNEL	1	2	5	6	8	9	13	14 OF 0-15

APPENDIX D

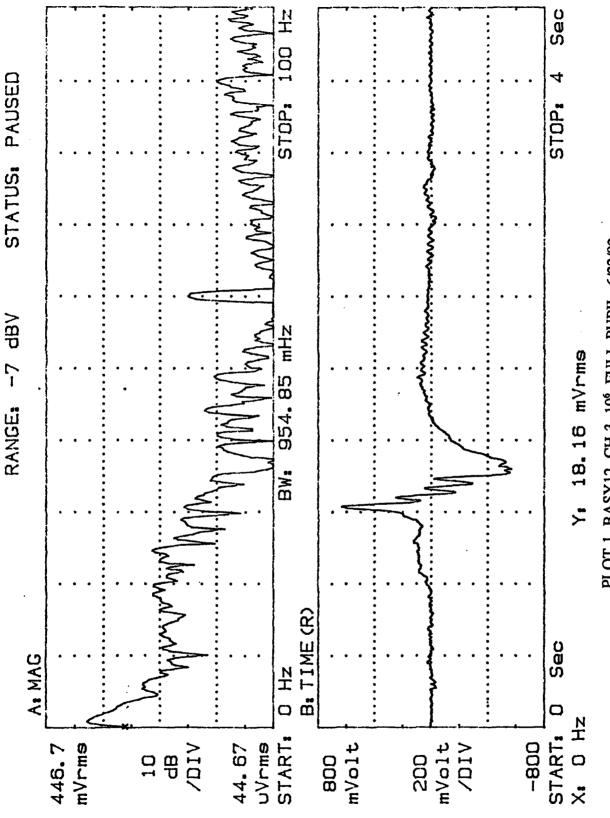
	WEA	THER CONDITIONS	
	6/22/92	6/23/92	6/24/92
	60% CLOUDS, 40% SUNSHINE	35% CLOUDS, 65% SUNSHINE	85% CLOUDS, 15% SUNSHINE
TIME	ATI	MOSPHERIC PRESSURE (m	m Hg)
6:00	760.48	762.25	756.16
12:00	760.73	761.49	755.65
18:00	760.48	758.70	753.36
	TEMPERATURE	(°F)/RELATIVE HUMIDITY	(%)
1:00	55/63	55/68	67/58
2:00	54/66	54/74	67/54
3:00	54/68	52/79	68/58
4:00	54/66	53/79	67/67
5:00	52/71	52/79	67/72
6:00	52/71	51/82	67/75
7:00	53/66	54/80	69/72
8:00	56/61	58/71	67/83
9:00	58/57	62/55	68/83
10:00	58/52	66/48	70/78
11:00	57/54	68/38	70/78
12:00	60/47	70/37	71/78
13:00	60/49	71/37	74/65
14:00	63/46	73/36	78/59
15:00	63/44	74/36	79/57
16:00	65/39	75/35	80/53
17:00	66/39	76/32	78/57
18:00	67/39	74/39	73/73
19:00	67/39	74/40	70/75
20:00	65/42	72/47	71/78
21:00	62/49	69/56	69/83

APPENDIX E

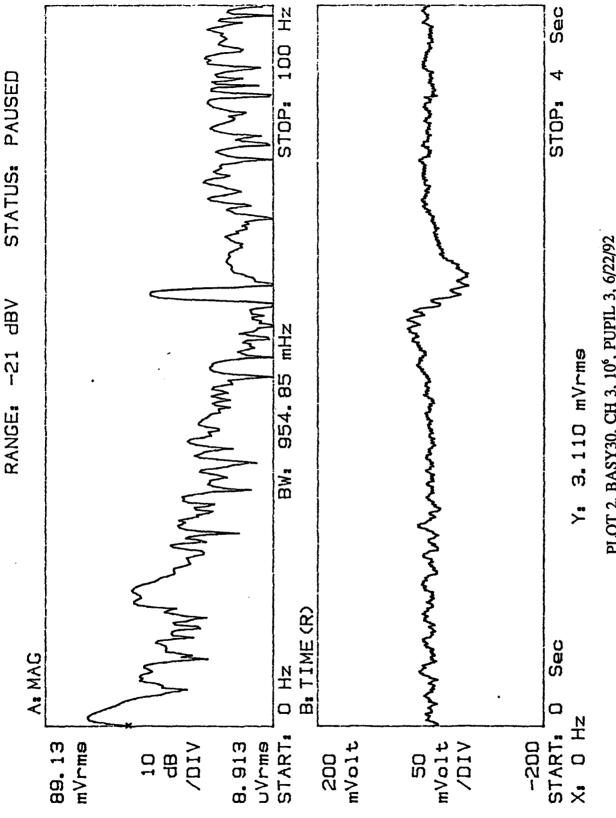
Section of Digitized Data File - BASY12.TST



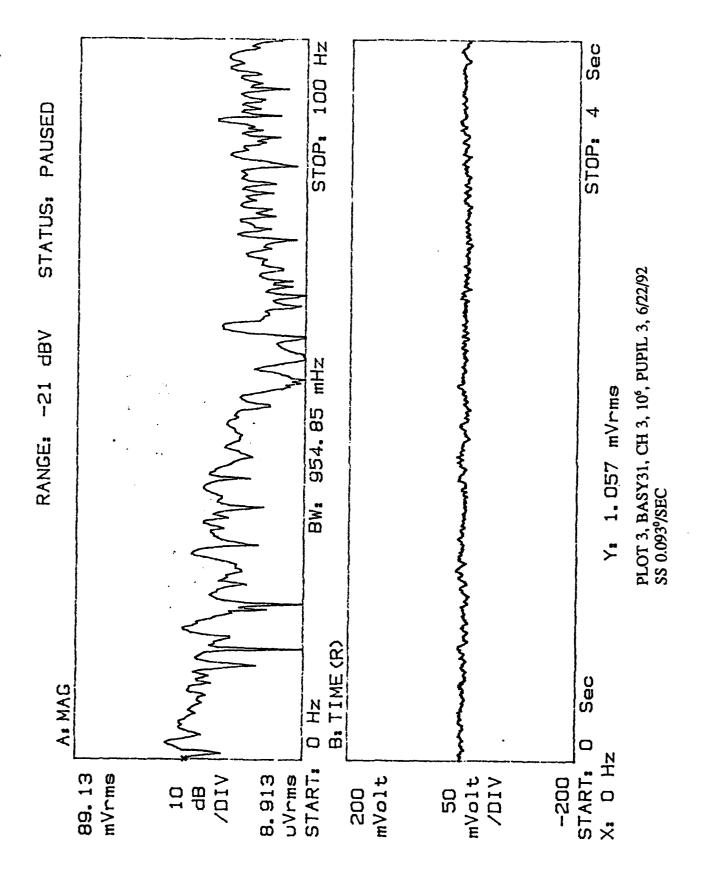
CH 3, 10°, FULL PUPIL, 6/22/92 SS 0.093/SEC, 600 C BLACKBODY AT 206 M

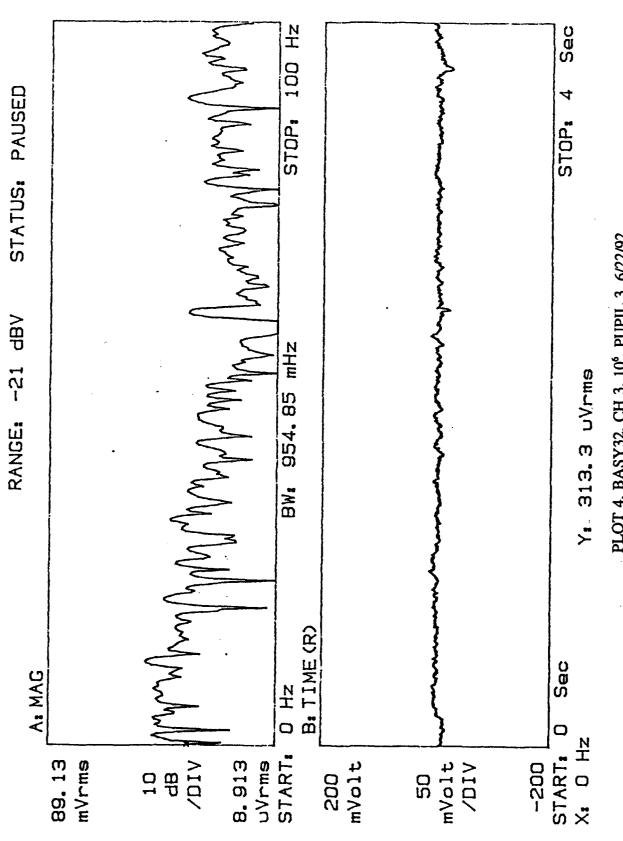


PLOT 1, BASY12, CH 3, 10°, FULL PUPIL, 6/22/92 SS 0.093°/SEC, 600 C BLACKBODY AT 206 M

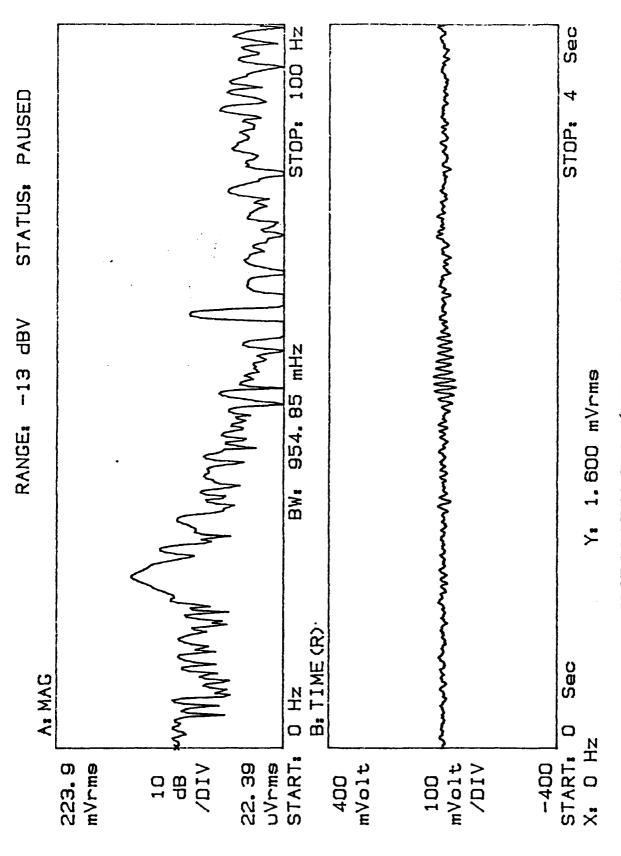


PLOT 2, BASY30, CH 3, 10°, PUPIL 3, 6/22/92 SS 0.093°/SEC, 375 C BLACKBODY AT 206 M

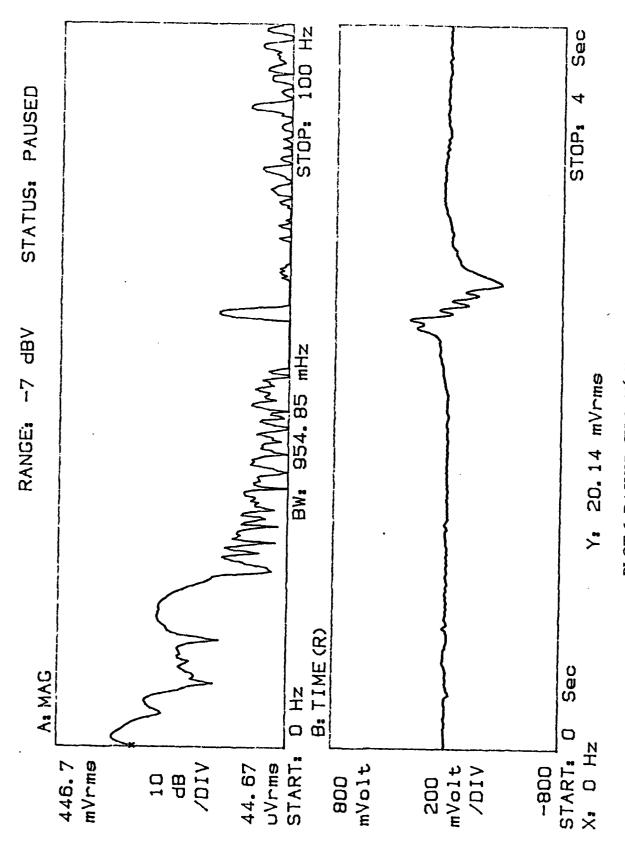




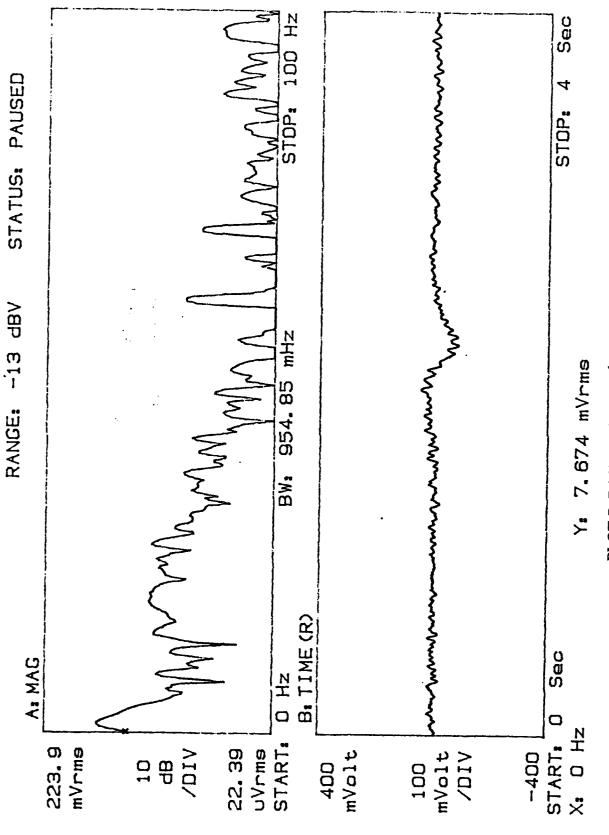
PLOT 4, BASY32, CH 3, 10°, PUPIL 3, 6/22/92 SS 0.093°/SEC, CIRRO-STRATUS AT > 5 KM



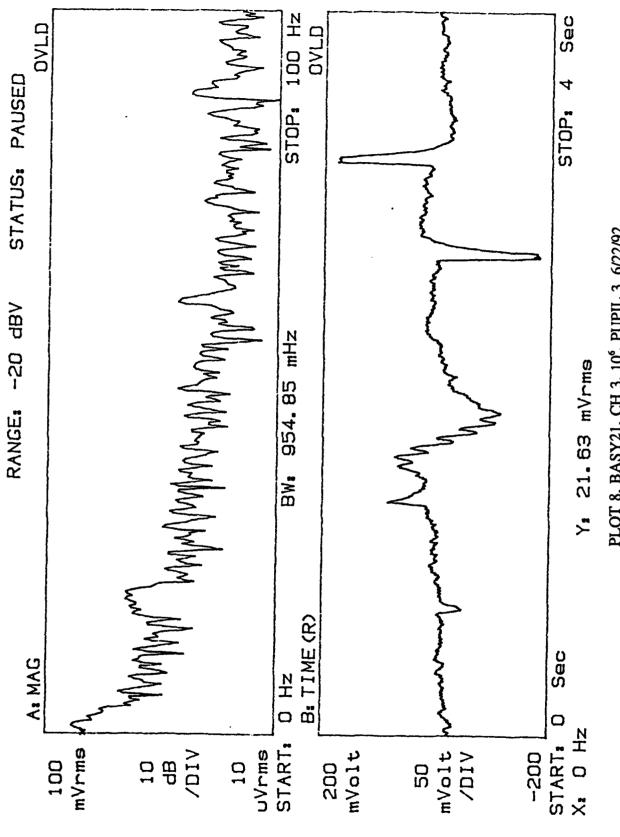
PLOT 5, BASY24, CH 3, 10°, FULL PUPIL, 6/22/92 NON-SCANNING CLOSE TO 375 C BLACKBODY AT 206 M



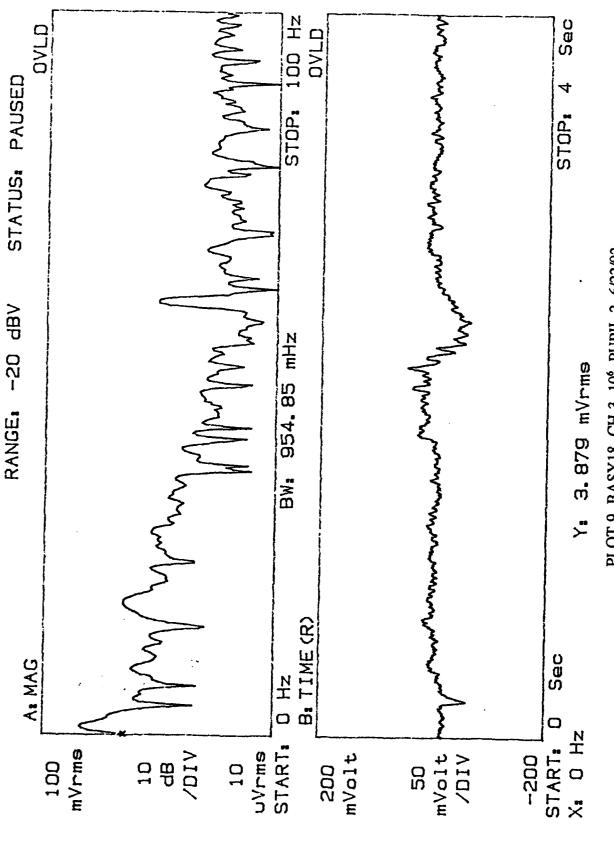
PLOT 6, BASY27, CH 3, 10°, FULL PUPIL, 6/22/92 SS 0.093°/SEC, 375 C BLACKBODY AT 206 M



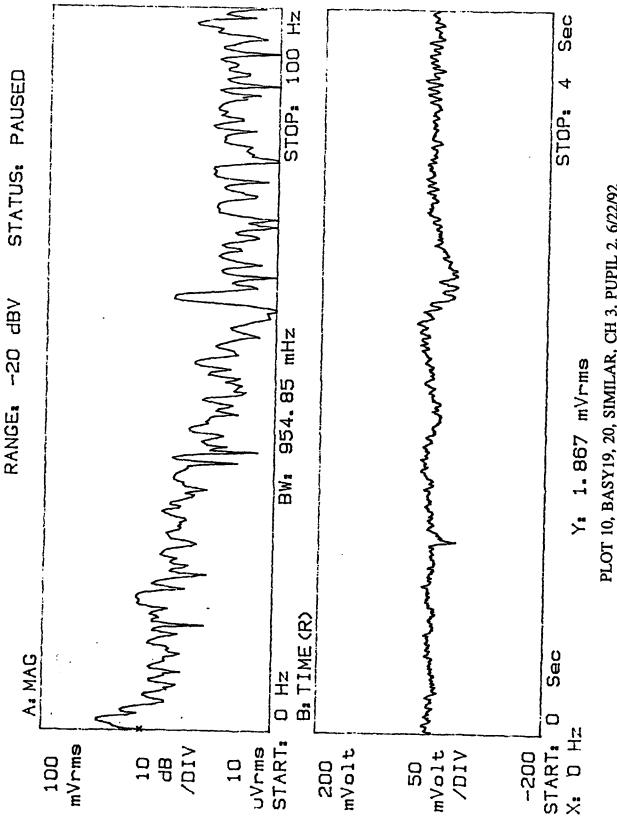
PLOT 7, BASY29, CH 3, 10°, PUPIL 3, 6/22/92 SS 0.093°/SEC, 375 C BLACKBODY AT 206 M



PLOT 8, BASY21, CH 3, 10°, PUPIL 3, 6/22/92 SS 0.093°/SEC, 500 C BLACKBODY AT 206 M

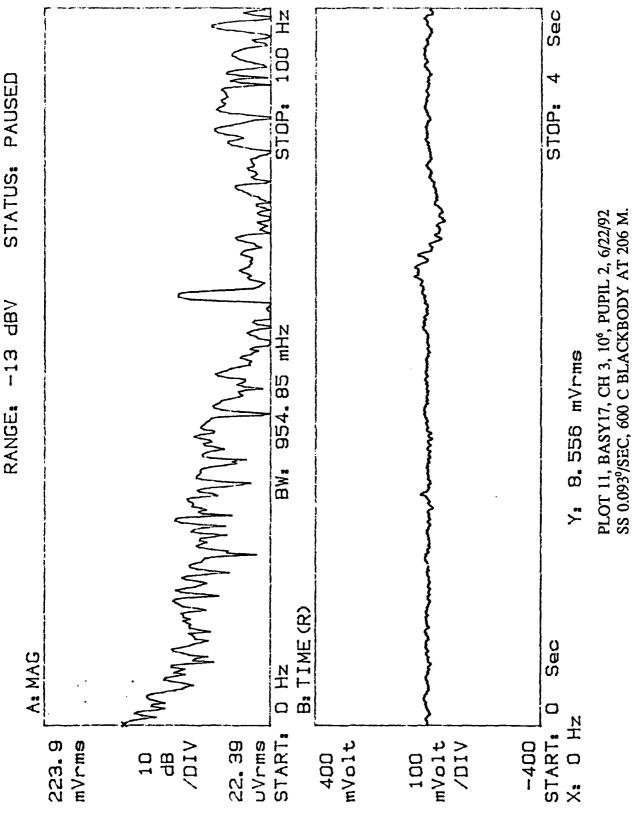


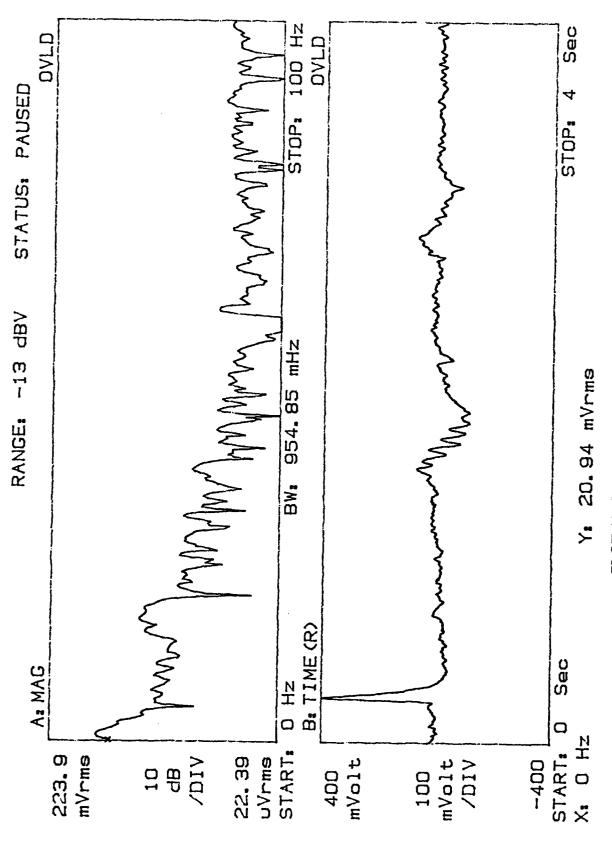
PLOT 9, BASY18, CH 3, 10°, PUPIL 2, 6/22/92 SS 0.093°/SEC, 600 C BLACKBODY AT 206 M



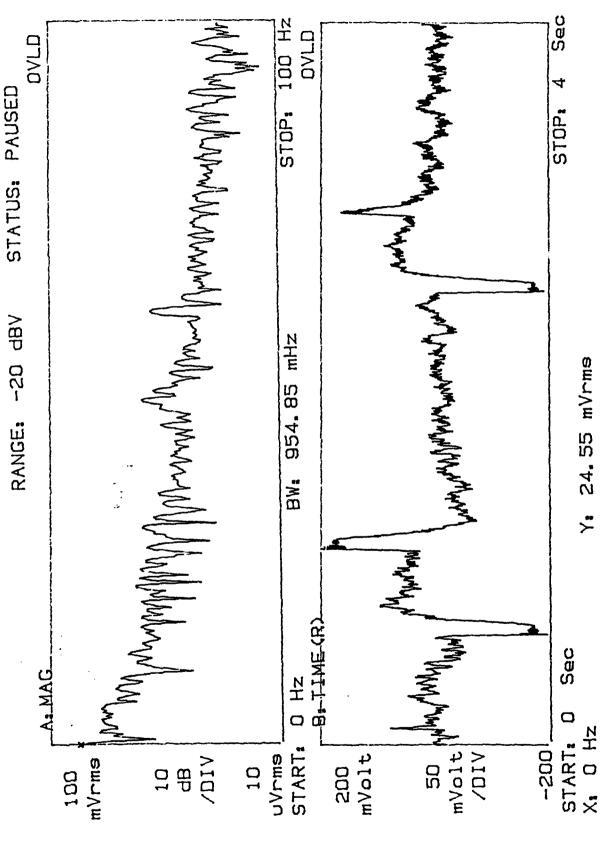
1

PLOT 10, BASY19, 20, SIMILAR, CH 3, PUPIL 2, 6/22/92 SS 0.093°/SEC, 600 C BLACKBODY AT 206 M





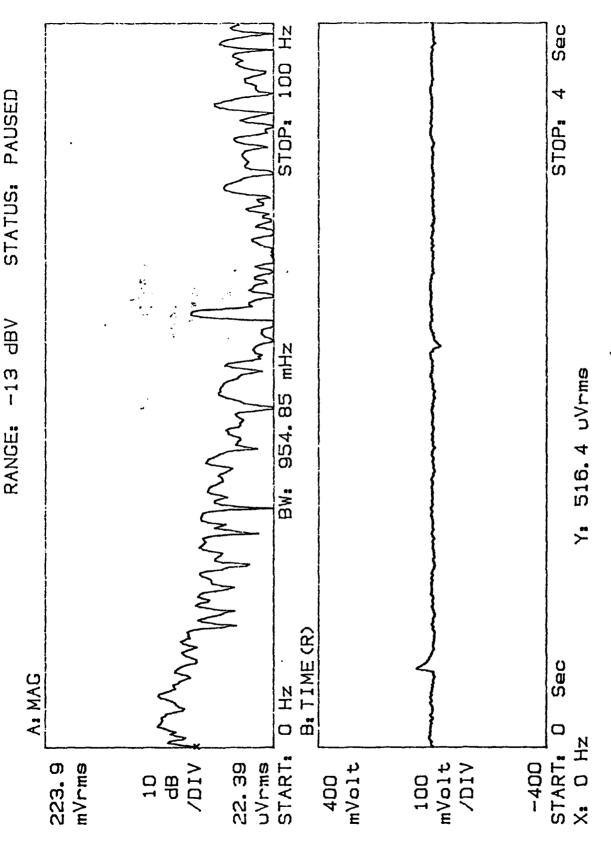
PLOT 12, BASY16, CH 3, 10°, PUPIL 3, 6/22/92 SS 0.093°/SEC, 600 C BLACKBODY AT 206 M



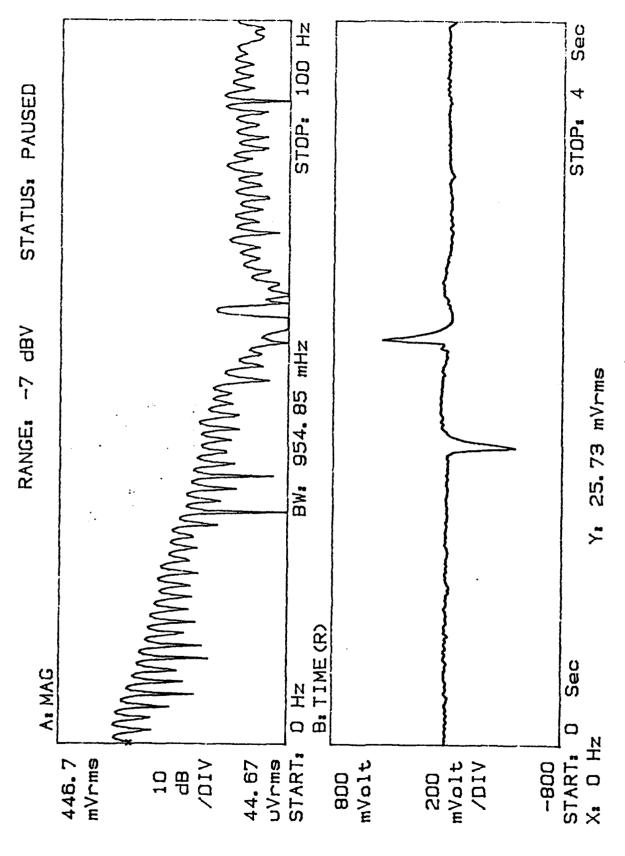
PLOT 13, NO FILE, CH 3, 10°, PUPIL 2, 6/22/92 SS 0.093°/SEC, APERTURE COVERED, TYPICAL TRANSIENTS

7

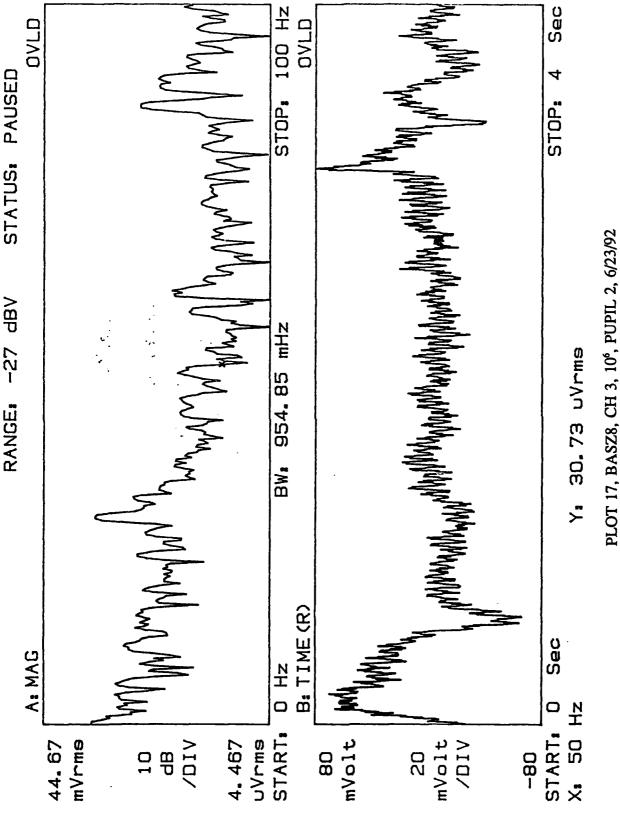
PLOT 14, BASY23, CH 3, FULL PUPIL, 6/22/92 SS 0.093°/SEC 500 C BLACKBODY AT 206 M



PLOT 15, BASZ25, CH 3, 10°, PUPIL 2, 6/22/92 NON-SCANNING, GRASS AT 206 M



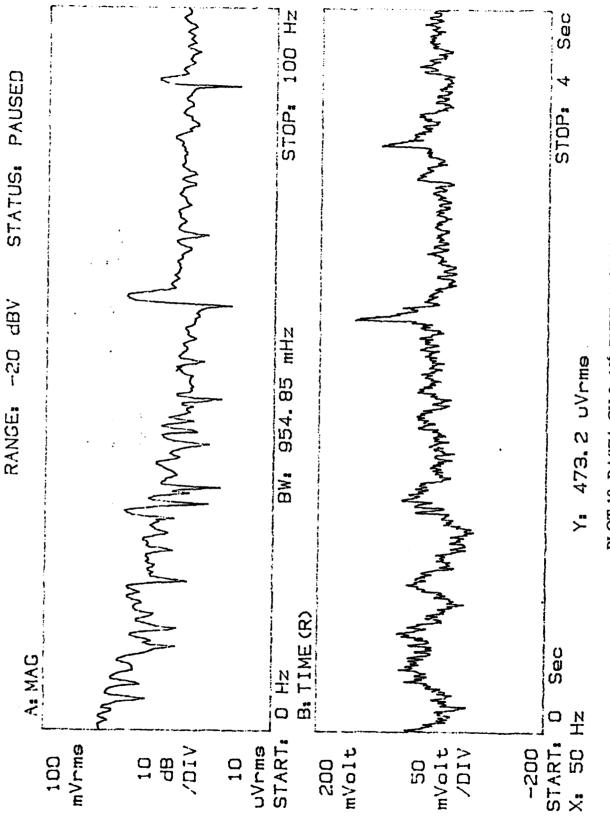
PLOT 16, NO FILE, CH 3, 10°, PUPIL 2, 6/22/92 NON-SCANNING, TYPICAL NOISE SPIKE



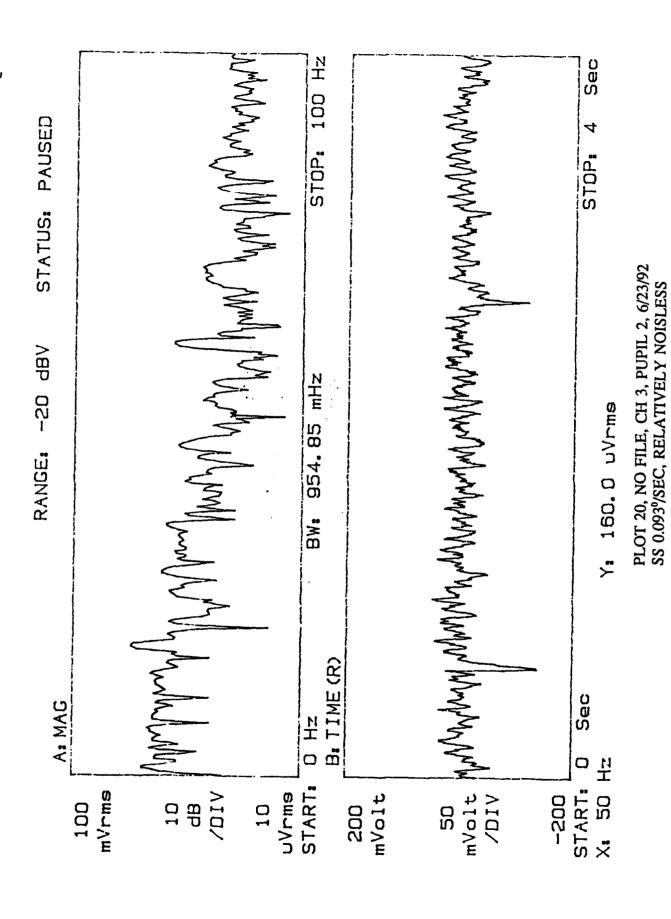
PLOT 17, BASZ8, CH 3, 10°, PUPIL 2, 6/23/92 NON-SCANNING, -20° AZ, +15° EL

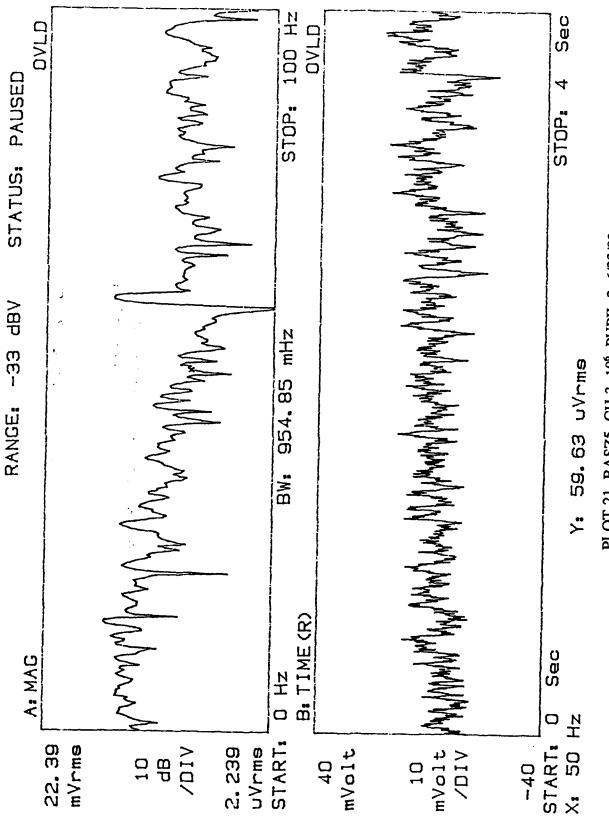
70

PLOT 18, BASZ7, CH 3, 10°, PUPIL 2, 6/23/92 SS 0.093°/SEC



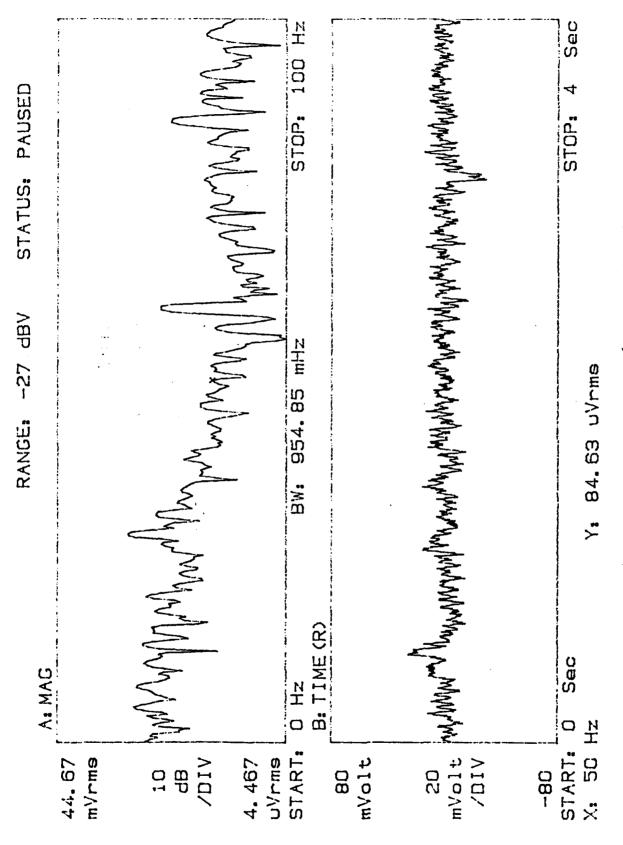
PLOT 19, BASZ4, CH 3, 10°, PUPIL 2, 6/23/92 SS 0.093°/SEC, BUILDING AT 200 M





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PLOT 21, BASZ5, CH 3, 10°, PUPIL 2, 6/23/92 SS 0.093°/SEC, RELATIVELY NOISELESS



PLOT 22, BASZ6, CH 3, 10°, PUPIL 2, 6/23/92 SS 0.093°/SEC, TOP-LIT CUMULOUS

